PH111: Introduction to Classical Mechanics Chapter 4: Non-Inertial Frames and Pseudo Forces

- Question: How do laws of physics change, when we change the frame of reference (coordinate system)?
- Are laws of physics same in all inertial frames of reference, i.e., frames moving with constant velocities?
- What if the frames of reference are accelerating?
- Underlying assumption will be that frames are moving with nonrelativistic velocities ($v \ll c$)
- For relativistic velocities, correct theory is Einstein's Special theory of relativity.

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Inertial Frames of Reference

- First we consider inertial frames of reference
- We demonstrate that in inertial frames of reference, Newton's second law holds good
- ullet Let lpha and eta be two frames of reference displaced by vector S



- We consider the dynamics of a particle of mass *m* in both the frames
- Position r_{β} of the particle in frame β , is related to its position r_{α} in frame α by

$$r_{\beta} = r_{\alpha} - S$$

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 If an observer in frame α measures the acceleration of mass m to be a_α, according to her the force acting on the mass F_α is

$$\mathsf{F}_{lpha}=m\mathsf{a}_{lpha}$$

• Similarly the observer in β on measuring its acceleration to be a_β will conclude that the force is

$$F_{\beta} = ma_{\beta}.$$

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• Question is: What is the relationship between F_{α} and F_{β} ?

• By taking time derivatives of $r_{\beta} = r_{\alpha} - S$, we obtain

$$\mathbf{v}_{eta} = \mathbf{v}_{lpha} - \mathbf{V}$$

 $\mathbf{a}_{eta} = \mathbf{a}_{lpha} - \mathbf{A}$

- Where $V = \dot{S}$ and $A = \ddot{S}$, are the velocity and acceleration, respectively, of frame β w.r.t. frame α .
- If A = 0, i.e., $m{eta}$ is an inertial frame, then

$$a_{\beta} = a_{\alpha}$$
$$\implies F_{\alpha} = ma_{\alpha} = ma_{\beta} = F_{\beta}$$

• Thus measured force is same in both the frames, as is the equation of motion.

Non-inertial Frames

- Thus, Newton's second law is unchanged when the frames of reference are inertial
- What about non-inertial frames of reference, i.e., when $A \neq 0$?
- We already have the result that

$$\begin{array}{l} \mathsf{a}_{\beta} = \mathsf{a}_{\alpha} - \mathsf{A} \\ \Longrightarrow \ \mathsf{F}_{\beta} = m \mathsf{a}_{\beta} = m \mathsf{a}_{\alpha} - m \mathsf{A} = \mathsf{F}_{\alpha} + \mathsf{F}_{p} \neq \mathsf{F}_{\alpha} \\ \text{where } \mathsf{F}_{p} = -m \mathsf{A}. \end{array}$$

Above notation F_p stands for pseudo Force.

• Thus force measured by an observer in a non-inertial frame is different from the one measured by an observer in an inertial frame.

Non-inertial frames and pseudo forces

- According to the observer in the non-inertial frame, the object is experiencing an additional force -mA, in a direction opposite to that of the acceleration
- Because this force is absent for an observer in the inertial frame, it is called "Pseudo Force".
- To illustrate this, we consider the example of a pendulum in an accelerating car

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Example: Pendulum in an Accelerating Car

• Consider a car with a pendulum inside, moving with an acceleration A



- We want to find the tension in the string T, and the angle θ the pendulum makes from the vertical
- We will analyze the problem both in the lab frame (static on ground) and the accelerated frame (moving with car)

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Analysis in Lab Frame (Attached to the Ground)

• Free body diagram in the lab frame is

$$\frac{\theta}{W} = \frac{T}{A}$$
Acceleration = A

- With respect to lab frame, mass *m* has an acceleration A
- As shown, it experiences two forces, tension T of the string, and its own weight W = mg
- We want to find angle of inclination heta, and au
- Application of Newton's law in vertical and horizontal directions, yields

 $T \cos \theta - W = 0 \text{ (vertical)}$ $T \sin \theta = mA \text{ (horizontal)}$

• Leading to the solution

$$an heta = rac{A}{g}$$
 $T = m \sqrt{g^2 + A^2}$

• Let us analyze the problem in the non-inertial frame, next

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Analysis in non-Inertial Frame (moving with the car)

• In the non-inertial (NI) frame, the free-body diagram is



- In the NI frame, particle is stationary, and in equilibrium
- But it is acted upon by three forces, instead of two
- Additional force is the fictitious (or pseudo) force $F_{fict} = -mA$
- Equations of motion are

$$-F_{fict} + T \sin \theta = 0 \text{ (horizontal)}$$
$$T \cos \theta - W = 0 \text{ (vertical)}$$

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- Because $F_{fict} = mA$, both these equations are essentially same as in case of inertial frame
- Thus we obtain the same expressions for T and θ .
- Next, we discuss different types of accelerating frames, i.e., the rotating frames of reference

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- Rotating frames of references are non-inertial
- Because any particle executing circular motion experiences centripetal acceleration
- Next, we develop the theory of rotating frames of references
- But, before that, we illustrate the vector nature of angular velocity

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Vector nature of angular velocity

• We denoted the position of a particle as a vector

$$\mathsf{r} = x\hat{\mathsf{i}} + y\hat{\mathsf{j}} + z\hat{\mathsf{k}}$$

• Can we similarly specify the angular position of a particle

$$\boldsymbol{\theta} = \theta_x \hat{\mathbf{i}} + \theta_y \hat{\mathbf{j}} + \theta_z \hat{\mathbf{k}}?$$

 The answer is no because such an expression does not satisfy commutative law of vector addition

$$\boldsymbol{\theta}_1 + \boldsymbol{\theta}_2 \neq \boldsymbol{\theta}_2 + \boldsymbol{\theta}_1$$

• Let us rotate a block first around the x axis, and then around the y axis. Compare that to the same operations performed in the reverse order

Non-commutative nature of finite rotations

• Consider those two rotations, with each one of them being $\pi/2$

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• Clearly $\theta_x \hat{i} + \theta_y \hat{j} \neq \theta_y \hat{j} + \theta_x \hat{i}$

Vector nature of Angular Velocity

• On the other hand, one can verify that infinitesimal rotations commute to first order terms

$$\Delta \theta_x \hat{i} + \Delta \theta_y \hat{j} \approx \Delta \theta_y \hat{j} + \Delta \theta_x \hat{i}$$

- Thus, infinitesimal rotations can be represented as vectors
- Because angular velocity is defined in terms of infinitesimal rotations

$$\boldsymbol{\omega} = \lim_{\Delta t \to 0} \frac{\Delta \boldsymbol{\theta}}{\Delta t},$$

• Angular velocity can be denoted as a vector

$$\boldsymbol{\omega} = \omega_x \hat{\mathbf{i}} + \omega_y \hat{\mathbf{j}} + \omega_z \hat{\mathbf{k}}$$

And, in general,

$$\boldsymbol{\omega} = \boldsymbol{\omega} \hat{\mathbf{n}},$$

where $\hat{\mathbf{n}}$ is the direction of the axis of rotation, and $\boldsymbol{\omega}$ is the magnitude of the angular velocity.

Relation between linear velocity and angular velocity

 $\bullet\,$ It is obvious that angular velocity $\pmb{\omega},$ will give rise to linear velocity v

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• What is the mathematical relation between the two?

Linear and Angular Velocities

• Consider a rigid body rotating with a uniform angular velocity $\pmb{\omega} = \pmb{\omega} \hat{\mathbf{n}}$ as shown



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Linear and angular velocity...

- Note that the position vector r of a particular point in the rigid body, precesses about the axis of rotation, and forms a cone with its tip at the origin
- ϕ is the constant angle between r and $\hat{\mathsf{n}}$
- During precession, r traces a circle of radius $r\sin\phi$
- Let $\Delta heta$ be the angle by which r rotates in time Δt



Relation between v and $\boldsymbol{\omega}$

 It is obvious from the figure that the magnitude of change in r, i.e., Δr is given by

$$|\Delta \mathbf{r}| = 2r \sin \phi \sin \frac{\Delta \theta}{2}$$

• For $\Delta t \rightarrow 0$, $\Delta \theta \rightarrow 0$, so that

$$|\Delta \mathbf{r}| = 2r \sin \phi \sin \frac{\Delta \theta}{2} \approx 2r \sin \phi \frac{\Delta \theta}{2} \approx r \sin \phi \Delta \theta$$

• Leading to (for $\Delta t
ightarrow 0$)

$$\left|\frac{\Delta \mathbf{r}}{\Delta t}\right| = r \sin \phi \frac{\Delta \theta}{\Delta t}$$
$$\implies \left|\frac{d\mathbf{r}}{dt}\right| = r \sin \phi \frac{d\theta}{dt} = \omega r \sin \phi$$

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• Obviously, for $\Delta t
ightarrow 0$, $rac{dr}{dt}$ is in tangential direction



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• Keeping the direction and magnitude of $\frac{dr}{dt}$ in mind, we conclude

$$\frac{d\mathbf{r}}{dt} = \mathbf{v} = \boldsymbol{\omega} \times \mathbf{r}$$

Rate of change of a general rotating vector

- Previous discussion was about when position vector r was precessing with a constant angular velocity ω, about the axis in direction n̂.
- But the same arguments will hold if, instead of r, some general vector A, was precessing with a constant angular velocity ω, about the axis in direction n̂.
- Then, we will have

$$\frac{dA}{dt} = \boldsymbol{\omega} \times A.$$

- This is a very important general relation about the rate of change of rotating vectors.
- Let A = v, then using above, we get the expression for acceleration of a rotating particle

 $\frac{d\mathbf{v}}{dt} = \mathbf{a} = \boldsymbol{\omega} \times \mathbf{v} = \boldsymbol{\omega} \times (\boldsymbol{\omega} \times \mathbf{r}) \text{ (centripetal acceleration)}$

Physics in Rotating Reference Frames

- Consider a general vector A which is changing with time
- When observed from an inertial frame, its rate of change is $\left(\frac{dA}{dt}\right)_{in}$.
- $\bullet\,$ Suppose we have a non-inertial frame of reference which is rotating with a constant angular velocity Ω
- What is the rate of change of A, i.e., $\left(\frac{dA}{dt}\right)_{rot}$, with respect to the rotating frame?

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- Let $\hat{i}, \hat{j}, \hat{k}$ be the basis vectors of the inertial frame
- And $\hat{i}', \hat{j}', \hat{k}'$ be the basis vectors of the rotating frame

Physics in rotating frames...

- This means that $\hat{i}', \hat{j}', \hat{k}'$ are rotating w.r.t. the inertial frame with angular velocity Ω .
- At a given point in time, A in two frames can be expressed as

$$\begin{split} \mathsf{A} &= A_x \hat{\mathsf{i}} + A_y \hat{\mathsf{j}} + A_z \hat{\mathsf{k}} \text{ (inertial)} \\ \mathsf{A} &= A_x' \hat{\mathsf{i}}' + A_y' \hat{\mathsf{j}}' + A_z' \hat{\mathsf{k}}' \text{ (rotating)} \end{split}$$

• Therefore, in inertial frame

$$\left(\frac{dA}{dt}\right)_{in} = \frac{dA_x}{dt}\hat{\mathbf{i}} + \frac{dA_y}{dt}\hat{\mathbf{j}} + \frac{dA_z}{dt}\hat{\mathbf{k}}$$

Physics in rotating frames contd.

- We can also compute the time derivative of the second expression of A
- Keeping in mind that not only the components of A, but the basis vectors $\hat{i}', \hat{j}', \hat{k}'$ are changing with time due to rotation
- Therefore, the rate of change of A will be

$$\left(\frac{dA}{dt}\right)_{in} = \frac{dA'_x}{dt}\hat{\mathbf{i}}' + \frac{dA'_y}{dt}\hat{\mathbf{j}}' + \frac{dA'_z}{dt}\hat{\mathbf{k}}' + A'_x\frac{d\hat{\mathbf{i}}'}{dt} + A'_y\frac{d\hat{\mathbf{j}}'}{dt} + A'_z\frac{d\hat{\mathbf{k}}'}{dt}$$

• Because vectors $\hat{i}', \hat{j}', \hat{k}'$ are rotating with angular velocity $\Omega,$ w.r.t. to the inertial frame

$$\frac{d\hat{i}'}{dt} = \mathbf{\Omega} \times \hat{i}'$$
$$\frac{d\hat{j}'}{dt} = \mathbf{\Omega} \times \hat{j}'$$
$$\frac{d\hat{k}'}{dt} = \mathbf{\Omega} \times \hat{k}'$$

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Rotating frames

So that

$$\left(\frac{dA}{dt}\right)_{in} = \frac{dA'_x}{dt}\hat{\mathbf{i}}' + \frac{dA'_y}{dt}\hat{\mathbf{j}}' + \frac{dA'_z}{dt}\hat{\mathbf{k}}' + \mathbf{\Omega} \times \left(A'_x\hat{\mathbf{i}}' + A'_y\hat{\mathbf{j}}' + A'_z\hat{\mathbf{k}}'\right)$$

• First three terms denote the rate of change of A as seen in the rotating frame, i.e.,

$$\frac{dA'_{x}}{dt}\hat{i}' + \frac{dA'_{y}}{dt}\hat{j}' + \frac{dA'_{z}}{dt}\hat{k}' = \left(\frac{dA}{dt}\right)_{rot}$$

Leading to

$$\left(\frac{dA}{dt}\right)_{in} = \left(\frac{dA}{dt}\right)_{rot} + \mathbf{\Omega} \times A$$

 Because A is a general vector, the previous formula can be symbolically expressed as

$$\left(rac{d}{dt}
ight)_{in} = \left(rac{d}{dt}
ight)_{rot} + \mathbf{\Omega} imes$$

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Velocity and acceleration in a rotating frame

• Taking A = r, we have

$$\left(\frac{d\mathbf{r}}{dt}\right)_{in} = \left(\frac{d\mathbf{r}}{dt}\right)_{rot} + \mathbf{\Omega} \times \mathbf{r}$$
$$\implies \mathbf{v}_{in} = \mathbf{v}_{rot} + \mathbf{\Omega} \times \mathbf{r}$$

• On taking $A = v_{in}$, we get

$$\begin{pmatrix} \frac{d\mathbf{v}_{in}}{dt} \end{pmatrix}_{in} = \left(\frac{d\mathbf{v}_{in}}{dt} \right)_{rot} + \mathbf{\Omega} \times \mathbf{v}_{in}$$

$$\Longrightarrow \left(\frac{d\mathbf{v}_{in}}{dt} \right)_{in} = \left(\frac{d}{dt} \right)_{rot} (\mathbf{v}_{rot} + \mathbf{\Omega} \times \mathbf{r}) + \mathbf{\Omega} \times (\mathbf{v}_{rot} + \mathbf{\Omega} \times \mathbf{r})$$

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Acceleration in Rotating Frame

Or

$$\begin{pmatrix} \frac{d\mathbf{v}_{in}}{dt} \end{pmatrix}_{in} = \left(\frac{d\mathbf{v}_{rot}}{dt} \right)_{rot} + \left(\frac{d(\mathbf{\Omega} \times \mathbf{r})}{dt} \right)_{rot} + \mathbf{\Omega} \times \mathbf{v}_{rot} + \mathbf{\Omega} \times (\mathbf{\Omega} \times \mathbf{r})$$
$$= \left(\frac{d\mathbf{v}_{rot}}{dt} \right)_{rot} + \mathbf{\Omega} \times \mathbf{v}_{rot} + \mathbf{\Omega} \times \mathbf{v}_{rot} + \mathbf{\Omega} \times (\mathbf{\Omega} \times \mathbf{r})$$
$$\implies \mathbf{a}_{in} = \mathbf{a}_{rot} + 2\mathbf{\Omega} \times \mathbf{v}_{rot} + \mathbf{\Omega} \times (\mathbf{\Omega} \times \mathbf{r})$$

Above, we used that condition that $\Omega=\mbox{constant},$ so that $\dot{\Omega}=0.$

• Thus, the acceleration as seen in the rotating frame is

$$a_{\textit{rot}} = a_{\textit{in}} - 2\mathbf{\Omega} \times v_{\textit{rot}} - \mathbf{\Omega} \times (\mathbf{\Omega} \times r)$$

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Pseudo Forces in Rotating Frames

 Multiplying the previous equation by m on both sides, and using notations F_{rot} = ma_{rot} and F = ma_{in}, we have

$$F_{rot} = F - 2m\mathbf{\Omega} \times v_{rot} - m\mathbf{\Omega} \times (\mathbf{\Omega} \times r)$$
$$= F + F_{coriolis} + F_{centrifugal}$$
$$= F + F_{fict}$$

 Where F is the real force acting on the particle, while F_{coriolis} and F_{centrifugal} are pseudo (fictitious) forces

$$F_{coriolis} = -2m\mathbf{\Omega} \times v_{rot}$$
$$F_{centrifugal} = -m\mathbf{\Omega} \times (\mathbf{\Omega} \times \mathbf{r})$$

Probing Centrifugal and Coriolis Forces

• Consider a circular plank (say a merry-go-round) rotating with an angular velocity $\mathbf{\Omega} = \Omega \hat{\mathbf{k}}$, with a mass *m* as shown



- Mass m is moving with a velocity v which is in the radial direction w.r.t. plank
- Location of the particle at a given instant is r, w.r.t. to rotating coordinate system
- What are the magnitudes and directions of pseudo forces?

• Centrifugal force can be computed as

$$\begin{aligned} \mathsf{F}_{centrifugal} &= -m\mathbf{\Omega} \times (\mathbf{\Omega} \times \mathsf{r}) \\ &= -m\Omega^2 r \hat{\mathsf{k}} \times (\hat{\mathsf{k}} \times \hat{\mathsf{r}}) \\ &= -m\Omega^2 r \hat{\mathsf{k}} \times \hat{\boldsymbol{\theta}} \\ &= -m\Omega^2 r (-\hat{\mathsf{r}}) \\ &= m\Omega^2 \mathsf{r}. \end{aligned}$$

Thus centrifugal force has the same magnitude as the centripetal force, but opposite direction, as expected of a pseudo force.

• Coriolis force exists only when the particle moves with respect to the rotating frame. Here

$$v_{rot} = v\hat{r}.$$

• Therefore,

$$F_{coriolis} = -2m\mathbf{\Omega} \times \mathbf{v}_{rot}$$
$$= -2m\Omega \mathbf{v}(\hat{\mathbf{k}} \times \hat{\mathbf{r}})$$
$$= -2m\Omega \mathbf{v}\,\hat{\boldsymbol{\theta}}$$

Coriolis and Centrifugal Forces

Thus, finally the direction of the forces



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Coriolis Force due to Rotation of Earth

• Earth's Angular Velocity in a Non-Intertial Frame



- Here x points to south, y to east, and z is radially outwards (vertically above from earth), and λ is latitude angle
- In this frame

$$\mathbf{\Omega} = -\Omega\cos\lambda\hat{i} + \Omega\sin\lambda\hat{k}$$

Coriolis Force on a Falling Object

 If a particle of mass m is falling vertically down, at a given instant with velocity v, then

$$v = -v\hat{k}$$

• Then Coriolis force on it due to Earth's rotation is

$$\mathsf{F}_{c} = -2m(\mathbf{\Omega} \times \mathbf{v}) = -2m\Omega \begin{vmatrix} \hat{\mathbf{i}} & \hat{\mathbf{j}} & \hat{\mathbf{k}} \\ -\cos\lambda & \mathbf{0} & \sin\lambda \\ \mathbf{0} & \mathbf{0} & -\mathbf{v} \end{vmatrix} = 2mv\Omega\cos\lambda\hat{\mathbf{j}}$$

- Thus, the object will experience a force towards east, and will get deviated in that direction
- Another example: away from equator, wind flow becomes circular due to Coriolis force
- Note $F_c \perp v_{rot}$, so it will lead to a circular motion

- The Foucault pendulum (FP) is a fine example of Coriolis Force
- It clearly demonstrates that we on earth are located in a rotating frame
- That is, a non-inertial frame
- Very good information on FP is available on Wikipedia with all the history
- Check out the Wikipedia page here
- The following simplified treatment is based on a solved example in Kleppner and Kolenkow.

- If consider a large enough pendulum, we will see that its plane of oscillations rotates
- That is the pendulum doesn't keep oscillating in the same plane
- This rotation and its period can be explained mathematically in terms of the Coriolis force

• Consider the following figure showing a pendulum on Earth's surface



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- ullet In the figure, λ denotes the latitude angle
- (r, θ) denote the plane polar coordinates of the bob of the pendulum
- the coordinate system is supposed to be attached to the earth
- the earth is supposed to be flat
- A good approximation given that bob doesn't move over large distances
- We further assume that the length of the pendulum is I
- Let us setup the equations of motion of the bob

- Clearly, the bob is moving along the r direction
- ullet If its motion were in a plane, $m{ heta}$ will not change with time
- Then $\dot{ heta}=0$
- ullet The frequency of oscillations w of the pendulum will be

$$\omega = \sqrt{\frac{l}{g}}$$

• Therefore, r(t) will be given by

$$r(t)=r_0\sin\omega t,$$

where r_0 is the amplitude

 \bullet Assume, as before, the angular velocity of the rotation of earth is $\pmb{\Omega}$

$$\mathbf{\Omega} = \Omega \sin \lambda \,\hat{k} - \Omega \cos \lambda \,\hat{r},$$

where \hat{k} denotes the perpendicular to earth's surface at the lattitude λ

• If the mass of the bob is *m*, the Coriolis force acting on the bob is

$$F_c = -2m(\mathbf{\Omega} \times v)$$

• Given that to a good approximation $\mathbf{v}=\dot{r}\hat{\mathbf{r}},$ we obtain from above

$$F_{c} = -2m(\Omega \sin \lambda \hat{k} + \Omega \cos \lambda \hat{r}) \times (\dot{r}\hat{r})$$
$$= -2m\dot{r}\Omega \sin \lambda \hat{\theta}$$

- Thus Coriolis force is strictly in the tangential $(\hat{oldsymbol{ heta}})$ direction
- So we set up the equation of motion in the tangential direction

$$m(2\dot{r}\dot{\theta}+r\ddot{ heta})=-2m\dot{r}\Omega\sin\lambda$$

Leading to

$$2\dot{r}\dot{\theta} + r\ddot{\theta} = -2\Omega\sin\lambda\dot{r}$$

- Although, this equation can be solved quite precisely, but we will use an approximate approach
- Reasonable to assume $\dot{\theta}$ (angular speed of precession) to be constant
- Implying that $\ddot{\theta} = 0$

• This leads to the equation

$$2\dot{r}\dot{ heta} = -2\Omega\sin\lambda\dot{r}$$

• Thus we obtain the solution of the problem

$$\dot{ heta} = -\Omega \sin \lambda$$

• With $|\dot{\theta}| = \Omega \sin \lambda$, we obtain the period of the precession to be

$$T = \frac{2\pi}{|\dot{\theta}|} = \frac{2\pi}{\Omega \sin \lambda} = \frac{24 \text{ hr}}{\sin \lambda}$$

- For Paris, $\lambda \approx 49^{\circ}$, so that $T_{Paris} \approx 32$ hr which is very close to the observed value of 31 hr 50 minutes!
- Clearly, at the north pole $T_{north-pole} \approx 24$ hr